Card 1 (AL4) January 1974 (AL4 October 1976) AP 101B-2305-14 Issue 3

FLIGHT REVERENCE CARDS

# JET PROVOGE T Mk 5 & 5A

# NORMAL DRILLS

# AIRCRAFT 'SAFE FOR PARKING'

The aircraft is safe for parking when safety pins are inserted in:

Both ejection seat face-screen firing handles Both ejection seat seat-pan firing handles Both ejection seat manual separation handles The canopy MDC handle (other pin stowed)

Prepared by Procurement Executive, Ministry of Defence, in collaboration with RAF Handling Squadron

BY COMMAND OF THE DEFENCE COUNCIL

#### NOTES TO USERS

- 1. These Flight Reference Cards are complementary to the Jet Provost T Mk 5 and 5A Aircrew Manual (AP 101B-2305-15). The same conventions are used and amendment procedure is similar.
- 2. Where necessary, separate checks are included for the T Mk 5 and T Mk 5A, the Mk 5A cards being identified by blue edges: other differences between the two marks of aircraft, eg limitations, are made clear on the appropriate cards.
- 3. Comments and suggestions regarding these Flight Reference Cards should be forwarded to the Officer Commanding, RAF Handling Squadron, Boscombe Down, Salisbury, SP4 0JF.

LIST OF CARDS (AL4)

Card No	Issued by	Card No	Issued by
1	AL4	12	AL3
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10	AL3	21	AL4
11	AL4		

<sup>\*</sup>Cards marked thus should incorporate manuscript amendments.



External Checks

## EXTERNAL CHECKS

Engine intake blanks removed, intakes clear Pitot head cover removed External surfaces undamaged Panels and filler caps secure Landing gear:

Ground locks removed
Oleo extensions equal and normal
Tyres free from cuts and creep
Brake leads secure, no hydraulic leaks
Static vent plugs removed
Jet pipe:

Cover removed
Free from wrinkling and distortion
Canopy and windscreen clean and undamaged

### COCKPIT CHECKS

Carry out the 'Safe for Parking' checks, then:
Canopy clutch ... Engaged
Hand fire extinguisher ... Secure, not discharged
Check for loose articles

### **Ejection Seat**

Main oxygen hose at seat break point ... Emergency oxygen bottle

Barometric time release unit ... ...

Face-screen firing cable

Canopy MDC gun sear ... ... Ejection gun sear ... ... Drogue withdrawal line

Top latch ... ...

Parachute withdrawal line

Drogue gun ... ...

Drogue gun static rod ... Mic-tel plug ... ... Guillotine firing unit sear

Manual separation handle Down and engaged

Manual separation handle PSP lanyard ... ...

Combined harness

Connected

Hose connected, safety pin removed, not fired

Static rod secured by quickrelease pin

Attached to ejection gun sear and to canopy disintegration gun sear

Safety pin removed Safety pin removed

Routed over drogue link line. Secured to drogue gun piston rod

Internal plunger flush with housing

Connected and routed through guillotine (behind yellow guard)

Safety pin removed Secured by quick-release pin Fully engaged

Safety pin removed; torque rod attachment connected

Down and engaged
Exposed length not beyond
red line

Tug the lap and shoulder straps to check for security

Note: For solo flying leave the right seat safety pins in position but have both canopy • 4 MDC firing unit safety pins removed. Check equipment restraining apron is securely fastened.

# TMk 5

### PRE-STARTING CHECKS

Cockpit checks/ Pre-starting Mk 5 Strap in Safety pins ... All stowed

Parking brake ... On

Landing gear ... DOWN buttons in Battery switch ... ON (cancel SWS) External supply ... On if required

UHF NORMAL/

STANDBY switch ... STANDBY

Flying controls ... Full and free movement with locking lever held up against

bottom catch. Adjust pedals to ensure that full rudder can

be applied

Cockpit Left Wall

Flap lever ... UP

Elevator trim control ... Full movement (no more than 300° each way) synchronised

with centre pedestal control.

Set neutral

Throttle ... ... Full and free movement,

synchronised with centre pedestal control. Set

CLOSED

Landing gear override ... Guarded central

### Below the Instrument Panel

Instrument dimming ... As required Cockpit lighting ... As required Anti-collision lights ... OFF

Cockpit emergency

lighting ... Test. OFF
Identification light ... As required

Taxy lamp/landing lamp

switch ... ... Central
Navigation lights ... As required

Oxygen (left and right) ... Connections. Supply wired on. Pressure 200 to 400 PSI. Air

inlet switch as required.
Flow indicator operating.
EMERGENCY — test for leaks, return switch to

central

# T Mk 5

1070	TO ADDRESS OF THE PARTY OF THE
Pre-starting Checks - con	tinued
Top temperature control	ON
Battery switch	ON .
Starting master switch	ON cancel SWS
Ignition switch	ON Cancer BWB
Pitot head heater	OFF
Landing gear emergency	011
lowering lever	Wired up
Flight instruments	When ap
START/OFF switch	START, artificial horizon and
Daring Cara Billion	altimeter OFF flags retract
Flight instruments	animotor off hage remot
NORMAL/STANDBY	
switch	NORMAL
Inverter test switch	
	Test, MI black, no OFF flags visible. Switch off, MI white,
	no OFF flags visible
Instrument Panels	in orr mage more
	a mine and a
Radio/radar lights dim-	A a magnified
ming	As required
Clock	Set
Fuel gauges	Check contents
UHF	T/R, volume, frequencies set
Standby altimeter	Set as required
Machmeter	Condition
Windscreen DE-ICE	Press ON button for 2 seconds
	and release. Check light
	ON/OFF
Landing gear position	Cl. 1. 1. Links and
indicator	Check three green lights and
The second second	bulb changeover, DAY / NIGHT screen as required
DDIC COLUMN	
RPM indicator	Condition
JPT indicator	Condition
Oil pressure gauge	Condition
Flap indicator	Note indication agrees with
ACT	flap position
ASI	Condition. UC warning flag
White the state of	not visible
Altimeter	Set to zero
Artificial horizon	Erect
Mk 4 Compass	Synchronise, check with E2B.
2 N 19 19 19 19 19 19 19 19 19 19 19 19 19	
Accelerometer	Condition
VSI Turn-and-slip indicator	Condition
Turn-and-slip indicator	OFF flag retracted
Turn-and-slip supply	NORMAL
Flight instrument panel	Secure
1.	(continued)

Pre-starting Mk 5

·(contd)

### TMk 5

# Pre-starting Checks — continued

Standard warning panel Test, night screen as required DME aerial sensitivity ... HIGH DME OFF **◆** Transponder SB. Code zero set DME indicator ... Condition Cockpit altimeter Condition Fuel tank air pressure ... Condition Brakes pressure gauge ... Condition Contents Oxygen ... Not discoloured Fire bottle indicator ... Emergency depressurisation selector ... Down Cockpit temperature selector ... As required Pressurisation ... UNPRESS Windscreen demist Push to off Rain/ice clearance OFF ••• Standby UHF ... Check A Standby UHF power ... NORMAL UHF NORMAL/ STANDBY switch ... NORMAL UHF aerial TOP Centre Pedestal Canopy seal Off Landing gear override ... Guarded central LP cock ... ON and gated. FUEL caption out 14 HP cock ... Test relight. OPEN and gated Throttle CLOSED Flap lever UP ... UHF mute switch Unmuted Aileron trim ... Full and correct movement Set neutral Canopy jettison/MDC handle ... Fully down Anti-collision lights ON

### TMk 5

### ENGINE STARTING

Starting Procedure

1. Press starter button for approximately 2 seconds and release

Light-up occurs at approximately 10% RPM.

2. If JPT increases rapidly through 650°C close HP cock, Allow engine to run down completely and repeat starting procedure using a different power supply.

3. If starting cycle cuts out prematurely, wait 40 seconds. Attempt

a further start using a different power supply.

4. If the starter indicator light remains on after RPM reach idling:

Post-mod 1772, shut down the engine and investigate. Pre-mod 1772, switch off the starter master switch, then:

a. If the light still remains on, set HP cock CLOSED, switch off battery master switch and have fault investi-

b. If the light goes out, the aircraft may be flown but have the fault investigated on landing.

Checks During Starting

JPT of T. . . ... 800°C (maximum)

RPM .... ... Increasing to 40 to 45%

... Rising Oil pressure

#### Failure to Start

If RPM stabilise at 15% (wet start):

1. Set HP cock CLOSED.

Ignition switch OFF. Carry out dry run. Drain

collector tanks and jet pipe.

Repeat start using a different power supply. After three starter cycle sequences allow starter to cool for 15 minutes and have fault investigated. If using internal batteries, only two starting attempts are permissible before the batteries must be changed.

Checks After Starting

Starter indicator light ... Out

External supply ... Disconnected

... 500°C (maximum) JPT

... 40 to 45% RPM ...

8 PSI (minimum) Oil pressure

Fuel valves, left and right Test, light out within 3 seconds.

If not, re-test when fuel air pressure stabilises at 1.5 to

2:5 PSI

Inverter MI Black

All captions out SWP

Transponder TEST

Green sector Hydraulic pressure

Wheelbrakes pressure ... 900 to 1000 PSI Flaps Test, leave UP

Airbrakes Test, select IN Card 5 (AL2)

# TMk 5 TAKE-OFF

S. A. D. L. S. E.			
Checks Before	Take-o	ff	7
TRIM		•••	Elevator neutral Aileron neutral
AIRBRAKES			IN, check visually
FUEL	90.		Contents sufficient
	22.0	•••	LP cock ON and gated
		\$0	HP cock OPEN and gated
			Fuel tank air pressure 1.5 to 2.5 PSI
FLAPS			TAKE-OFF
			1
INSTRUMENT	S		
Taxy light			ON
Pitot head he	ater		ON
Instruments		•••	Functioning, OFF flags re-
VII.2 (0.			tracted
Altimeters			Sub-scale set: within limits
Inverter MI		***	Black
DME			Standby, channel selected
Transponder			SB, code set
OXYGEN			Contents, connections, flow
HOOD			Select CLOSE
Canopy seal		•••	ON
Pressurisation			Select PRESS
HARNESS			Tight and locked
HYDRAULICS			Green sector
CONTROLS			Full, free and correct move-
			ment
CAPTIONS		•••	All out
PINS			All stowed
Checks During	Toko o	æ	
Control of the contro			Y 11: 00 of D D 6
Wheelbrakes			Holding at 90% RPM
IDT		•••	100 ± 0.5%
Oil programs	•••	•••	715°C (maximum) 20 PSI (minimum)
Oil pressure		•••	20 PSI (minimum)
Checks After T	ake-of	f	
Wheelbrakes			On/off
Landing gear			UP, lights out
			UP above 110 knots and above
70 11			100 feet above ground level
RPM		•••	101.5% (maximum)
JPT		varr	735°C (maximum)

Starting/ Take-off T Mk 5

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### PRE-STARTING CHECKS

Strap in Safety pins ... All stowed

Parking brake ... On

Landing gear ... DOWN buttons in Battery switch ... ON (cancel SWS)
External supply ... On if required
CCU ...

NORM/FAIL switch NORM

NORM/FAIL switch

✓ NORM/EMERG

switch ... NORM

Rotary switch ... UHF SBY Toggle switch ... UHF SBY

Standby UHF power

switch ... ... NORMAL

Flying controls ... Full and free movement with locking lever held up against bottom catch. Adjust pedals to ensure that full rudder

can be applied

Cockpit Left Wall

Flap lever ... UP

Elevator trim control ... Full movement (no more than

300° each way) synchronised with centre pedestal

control. Set neutral

Throttle ... Full and free movement, synchronised with centre

pedestal control. Set closed

Landing gear override ... Guarded central

Below the Instrument Panel

Instrument dimming ... As required Cockpit lighting ... As required

Anti-collision lights ... OFF

Cockpit emergency light-

ing ... ... OFF avigation lights ... As required

Navigation lights ... As a Taxy lamps ... OFF

Identification lights

switch ... Central

# Pre-Starting Checks — continued

RAD/INST DIMMER Oxygen (left and right) As required Connections. Supply wired on.

Pressure 200 to 400 PSI. Air inlet switch as required. Flow indicator operating. EMERGENCY — test for leaks, return switch to-

central

ON Top temp control Battery switch ... ON

Starting master switch ... ON (Cancel SWS)

Ignition switch ... ON Pitot head heater OFF

Landing gear emergency

lowering lever Wired up Flight instrument start

switch ... Switch to START; check artificial horizon, altimeter

and compass OFF flags retract

Flight instrument switch NORMAL

Inverter test Test: MI black, no OFF flags visible. Switch OFF, MI

white, no OFF flags visible

# **Instrument Panels**

Flap indicator ... Indication agrees with flap position

Landing gear position indication ...

Check three green lights and bulb changeover, DAY /

NIGHT screen as required

V/UHF ... T/R, frequency set

NAV controller OFF DME controller OFF JPT indicator ... Condition

(continued)

Pre-Starting Mk 5A

# Pre-Starting Checks - continued

Fuel tank switches Check contents, leave on WING and central Turn-and-slip indicator ... OFF flag retracted Standby artificial horizon OFF flag retracted Condition RPM indicator ... Oil pressure gauge Condition CCU STANDBY UHF Test Rotary switch COMM SBY toggle UHF Off switch COMM toggle switch On S/B. Code zero set Transponder ASI Condition, UC warning flag not visible Artificial horizon Erect, OFF flag retracted Altimeter OFF flag retracted; set as required Flight instrument panel Secure VSI Condition Windscreen DE-ICE Press ON button for 2 seconds and release. Check light ON/OFF COMP/DG switch COMP CL6 compass ... Synchronised. Check against E2B E2B compass light OFF switch ... Accelerometer ... Condition ... **◆** DME indicator ... Condition ... Marker switches ON; test marker lights; set OFF

Pre-starting

Mk 5A

(contd)

## T Mk 5A

### Pre-starting Checks — continued

#### ATTITUDE NORMAL/ STBY switch NORMAL ILS indicator ... Failure flags showing SWS Test, night screen as required Condition Machmeter Set as required **♦** Standby altimeter ... Cockpit altimeter Condition Clock Set Oxygen Contents Brakes pressure gauge ... Condition Fuel tank air pressure Condition Fire bottle indicator Not discoloured Emergency depressurisation selector Down Cockpit temperature selector As required Pressurisation ... UNPRESS ... Demist ... OFF Rain/ice switch OFF STANDBY GUARD/A CHAN-NEL switch ... GUARD UHF aerial switch TOP

#### Centre Pedestal

Canopy seal		Off
Landing gear over	ride	
switch	****	Guarded central
LP cock		ON and gated. FUEL caption
4 LP LOCK GUEL "CONH	m' bu	Test relight. OPEN and gated
HP cock		Test relight. OPEN and gated
Throttle		CLOSED
Flap lever	•••	UP
UHF mute switch		Unmuted
Aileron trim		Full and correct movement, set neutral
<b>♦</b> Canopy jettison/MD	C	
handle		Fully down
Anti-collision lights		ON

### ENGINE STARTING

#### Starting Procedure

 Press starter button for approximately 2 seconds and release. Light-up occurs at approximately 10% RPM.

 If JPT increases rapidly through 650°C close HP cock. Allow engine to run down completely and repeat starting procedure using a different power supply.

 If starting cycle cuts out prematurely, wait 40 seconds. Attempt a further start using a different power supply.

44. If the starter indicator light remains on after RPM reach idling:

Post-mod 1772, shut down the engine and investigate.

Pre-mod 1772, switch off the starter master switch, then:

a. If the light still remains on, set HP cock CLOSED, switch off battery master switch and have fault investigated.

b. If the light goes out, the aircraft may be flown but have the fault investigated on landing.

#### **Checks During Starting**

JPT ... 800°C (max)

RPM ... Increasing to 40 to 45%

Oil pressure ... Rising

#### Failure to Start

If RPM stabilise at 15% (wet start):

1. Set HP cock CLOSED.

2. Ignition switch OFF. Carry out dry run. Drain collector

tanks and jet pipe.

 Repeat start using a different power supply. After three starter cycle sequences allow starter to cool for 15 minutes and have fault investigated. If using internal batteries, only two starting attempts are permissible before the batteries must be changed.

### Checks After Starting

Starter indicator light ... Out External supply Disconnected JPT 500°C (max) ... RPM 40 to 45% 8 PSI (min) Oil pressure ◀ Transponder Test ... DME controller ... On, test NAV controller ... As required

Fuel valves, left and right

Test, light out within 3 sec.

If not, re-test when fuel air
pressure stabilises at 1.5 to

Inverter MI ... 2.5 PSI
Black

SWP ... ... ... All captions out
Hydraulic pressure ... Green sector
Wheelbrake pressure ... 900 to 1000 PSI
Flaps ... Test, leave UP
Airbrakes ... Test, select IN

### TAKE-OFF

### Checks Before Take-off

TRIM ... ... Elevator neutral

AIRBRAKES ... IN, check visually FUEL ... ... Contents sufficient

LP cock ON and gated HP cock OPEN and gated Fuel tank air pressure 1.5 to

2.5 PSI

FLAPS ... TAKE-OFF

## **INSTRUMENTS**

Taxy light ... ON Pitot head heater ... ON

Transponder ... S/B, code set NAV/DME ... As required

Instruments ... Functioning, OFF flags retracted

Inverter MI ... Black

Altimeters ... Sub-scale set: within limits OXYGEN ... Contents, connections, flow

HOOD ... Select CLOSE

Canopy seal ... ON

Pressurisation ... Select PRESS
HARNESS ... Tight and locked
HYDRAULICS ... Green sector

CONTROLS ... Full, free and correct move-

ment

CAPTIONS ... All out
PINS ... All stowed

# **Checks During Take-off**

Wheelbrakes ... Holding at 90% RPM

RPM ... ... ... 100 ± 0·5 % JPT ... ... 715 °C (max) Oil pressure ... 20 PSI (min)

#### Checks After Take-off

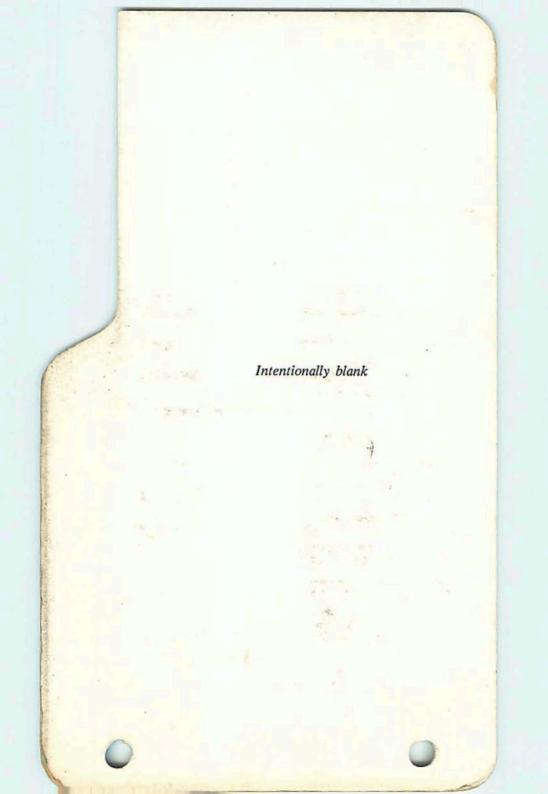
Wheelbrakes ... On/off

Landing gear ... UP, lights out

Flaps ... ... UP above 110 knots and above

RPM ... ... ...101.5% (max) JPT ... ... 735°C (max)

Starting/ Take-off Mk 5A



#### IN FLIGHT

### Pre-Stalling, Spinning and Aerobatic Checks

Height ... Sufficient for recovery (see relevant Air Staff Instructions)

Airframe ... Flaps and landing gear as required. Airbrakes test

Security ... Harness tight and locked.
Check for loose articles

Engine ... ... JPT, oil pressure and fuel state within limits

Location ... Clear of controlled air space and populated areas

Lookout ... Clear of other aircraft and cloud, vertically and horizontally

### Pre-Joining/Descent Checks

Fuel ... Contents and balance

Instruments ... Functioning, erect and synchronised

Radio ... ... Unmuted. Channel selected.

Altimeters ... Set as required, cross-checked

Demist ... ... As required

Take-Off/ In Flight

SPINNING ! ADDITIONAL CHECKS

NOT OVER 8/8 CLOUD OR SEA

WAL DEFINED HORIZON

CANOPY: CLEAR OF MIST AND ICE

FUEL BETWEEN 500 16 and 1780 16.

FUR IMBALANCE LESS THAN 100 16

## T Mk 5

### APPROACH

## Instrument Approach Settings

Position	Configuration	RPM%	Speed kt
Initial descent	A/B out	70	200
Slow rate descent	A/B out	70	140
Pattern and base leg	Gear down, flap T/O	75	Reducing to
Glidepath	Gear down, full flap	75	115

# Aircraft Approach Limitations (feet true)

Runway Aids Without Glidepath Guidance	GCA   PAR
250	200



To obtain indicated height, add 50 feet

### LANDING

# **Pre-Landing Checks**

SPEED ... ... Below 140 knots

AIRBRAKES ... In

LANDING GEAR

FUEL ... DOWN; three green lights

Sufficient, calculate threshold speed

FLAPS ... ... As required HARNESS ... Tight and locked

WHEELBRAKES ... Pressure in green sector. Operation, ensure pressure exhausted

# Circuit Speeds

	Circuit	Final Turn	A pproach	Threshold
Powered approach	140	115	Reducing	95
Glide approach	reducing	115	to threshold	100
Flapless approach	115	115	speed	105

Increase threshold speed by 5 knots if fuel 1500 lb or more. Reduce threshold speed by 5 knots if fuel 500 lb or less.

## TMk 5

### Landing - continued

### Checks After Landing

Parking brake ... On, hydraulic pressure in Green sector Flaps UP Taxy light OFF Pitot head heater OFF Rain/ice clearance OFF Windscreen demist OFF Pressurisation ... Select UNPRESS Canopy seal Transponder OFF. Set Code zero ... DME ... OFF SWS Test Hydraulic pressure Green sector

### SHUT-DOWN

On

#### Shut-Down Procedure

Parking brake ...

Chocks ...

position

... CLOSED Throttle ... Flaps ... TAKE-OFF HP cock ... CLOSED (when RPM stabi-... ... lised) LP cock ... ... OFF Flight instrument start ... OFF switch Ignition switch ... OFF Starting master switch ... OFF All electrical services ... OFF Battery switch ... OFF Flying controls ... LOCKED

... In position

Before unstrapping, have the aircraft made 'Safe for Parking'. When unstrapped, return the QRB to the locked

Approach/ Landings/ Shut-Down Mk 5

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### APPROACH

# **Instrument Approach Settings**

Position	Configuration	RPM%	Speed kt	
■ Initial descent	A/B out	70	200 ▶	
Slow rate descent	A/B out	70	140	
Pattern and base leg	Gear down, flap T/O	75	Reducing to	
Glidepath	Gear down, full flap	75	115	

# Aircraft Approach Limitations (feet true)

	or had a second	Runway Aids Without Glidepath Guidance	ILS	PAR
	In-line localiser	250	250	200
4	Offset localiser	270	270	- >

To obtain indicated height, add 50 feet

#### LANDING

# **Pre-Landing Checks**

SPEED ... Below 140 knots

AIRBRAKES ... In

LANDING GEAR ... DOWN; three green lights FUEL ... Sufficient, calculate threshold

speed

FLAPS ... ... As required HARNESS ... Tight and locked

WHEELBRAKES ... Pressure in green sector. Oper-

ation, ensure pressure

exhausted

# Circuit Speeds

	Circuit	Final Turn	Approach	Threshold
Powered approach	140	115	Reducing	95
Glide approach	reducing	115	to threshold	100
Flapless approach	115	115	speed	105

Increase threshold speed by 5 knots if fuel 1500 lb or more. Reduce threshold speed by 5 knots if fuel 500 lb or less.

## Landing - continued

# Checks After Landing

Parking brake ... On, hydraulic pressure in green sector

Flaps ... ... UP
Taxy light ... OFF
Pitot head heater ... OFF
NAV/DME ... OFF

Transponder ... OFF. Set code zero

Marker switches ... OFF Rain/ice clearance ... OFF Windscreen demist ... OFF

Pressurisation ... Select UNPRESS

Canopy seal ... ... OFF SWS ... ... Test

Hydraulic pressure ... Green sector

### SHUT-DOWN

#### Shut-Down Procedure

Parking brake ... ... On
Throttle ... ... CLOSED
Flaps ... ... TAKE-OFF

HP cock ... ... CLOSED (when RPM stabi-

lised)

LP cock ... ... Flight instrument start

switch ... ... OFF
Ignition switch ... OFF
Starting master switch ... OFF

All electrical services ... OFF
Radio ... ... OFF

CCU:
I/C VOL switch ... OFF
Toggle switches ... OFF

Battery switch ... OFF
Flying controls ... LOCKED
Chocks ... ... In position

Before unstrapping, have the aircraft made 'Safe for Parking'. When unstrapped, return the QRB to the locked position

Approach/ Landing/ Shut-Down Mk 5A

### AIRCRAFT LIMITATIONS

Note: Details in brackets apply to T Mk 5 aircraft which are fitted with tip tanks.

# Maximum Airpseeds

Clean aircraft			400 knots
	- Above 10,000 feet		0.73M
Canopy open			140 knots
Flaps	-Between UP and T/O, at 7	r/O	155 knots
ALTERNATION OF THE PARTY OF THE	-Between T/O and full, at f	ull	125 knots
Landing gear	-Lowering and locked do	wn	140 knots
	D 400		125 knots

### Maximum Weights

Take-off and emergency l	anding	8150 lb	(9000 lb)
Normal landing		8000 lb	(8000 lb)

# 

The aircraft is cleared for aerobatics at fuel states up to 2100 lb but the tip tanks must be empty and the CG must be forward of 19.5 inches AOD.

# Spinning

Inverted spinning is prohibited. Erect spins up to eight (four) turns permitted, providing fuel load 500 to 1780 lb (1300 lb) and fuel asymmetry does not exceed 100 lb (No fuel in the tip tanks.) Spinning in Mk 5 aircraft when flown solo is not permitted.

# **G** Limits

Positive				6	g
Negative-	-up to 350	knots	minus	2.5g (mi	nus 2·0g)
· -	-350 to 400	knots		m	inus 1.5g
(With tip	tanks, CG	between	19.5 and	1 20·4 inc	ches AOD,
					+3g and
	d to gentle				
	age pannie				
	e between	0.2g and	the appr	ropriate 1	negative g
limitation:					

30 seconds above 30,000 feet

25 seconds between 20,000 to 30,000 feet

20 seconds between 10,000 to 20,000 feet

18 seconds between 0 to 10,000 feet

Fuel booster pump must be operating

Allow 45 seconds between prolonged applications of neg g

Card 12 (AL3)

### Aircraft Limitations - continued

### Crosswind - T/O and Landing

Dry runway ... ... ... ... ... 30 knots Wet runway ... ... ... ... 25 knots

### Rolling Manoeuvres

Full aileron may be used between +4g and minus 1.5g (+4g and 0g up to 350 knots)

Outside these limits not more than ½ aileron is to be used

### Aircraft Arresting Gear

SPRAG PUAG Mk 21 CHAG BLISS BAK 9, 12 RHAG Mk 1 and 500 S

No damage will occur with trampling speeds 10 knots or below

No hazard exists at any speed

# **♦** Aircraft Arresting Barriers

The aircraft are cleared for use with RAF barriers Mk 5, 6, and 12, and 12A at 'light' setting.

### ENGINE LIMITATIONS

# Viper Mk 202 Engine

Por	ver Rating	Engine RPM %	Max JPT °C	Time limit
105	take-off	100±0-5	715	20 :
Max	in flight	101.5 max *	735	20 minutes
Intern	nediate	98 max	695	30 minutes
Max	cont	95 max	655	
Grou	nd idling	42.5 ± 2.5	500†	3 2

Note: Within the associated time limit, the overriding operating limitation is the one first obtained at that power rating, ie either RPM or JPT.

\* Increase in full throttle RPM is caused by governor creep. The amount and duration of any overspeed in excess of 101.5% must be recorded.

† Max permissible JPT during starting and relighting is 800°C. If, during starting, the JPT accelerates rapidly through 650°C, the HP cock must be closed immediately. Report the maximum temperature reached and the length of time above 800°C.

# Limitations

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# Oil Pressure

Minimum at ground idling RPM ... ... 8 PSI
Minimum at max continuous RPM ... ... 20 PSI
Normal at max continuous RPM ... ... 30 PSI

# OPERATING DATA

Height (ft)	Fuel Used (lb)	Distance (NM)	· Time (minutes)
SL to 5000	68	5	1.6
SL to 10,000	140	12	3.4
SL to 15,000	216	21	5.7
SL to 20,000	288	31	8.1
SL to 25,000	368	44	11.3
SL to 30,000	460	62	15.7
SL to 35,000	606	97	24.8

Climb speed 200 knots from SL until M=0.4: 0.4 M thereafter. Full power for 20 minutes from start of take-off, then max intermediate.

# CRUISE DATA

# Best Range

Ht (1000 ft)	SL	5	10	15	20	25	30	35
IAS (kt)	210	195	185	180	170	165	160	155
TAS (kt)	210	215	215	225	235	245	260	275
Fuel lb/min	22.7	19-9	17.1	15.8	14.4	13.6	13.1	12.8
ANM/100 lb	15.6	18.2	20.9	23.9	26.9	29.7	33.0	35.9

# 95% Range

Ht (1000 ft)	SL	5	10	15	- 20	25	30	35
IAS (kt)	250	235	225	220	210	205	190	180
TAS (kt)	250	255	265	275	285	300	310	315
Fuel lb/min	28.3	24.5	22.1	20.3	18.7	17-9	16.6	15-5
ANM/100 lb	14.8	17.4	19-9	22.5	25.6	28-2	31-4	34-1

# Endurance

Ht (1000 ft)	SL	5	10	15	20	25	30	35
IAS (kt)	130	125	125	125	125	125	130	130
TAS (kt)	135	140	145	155	170	190	210	230
Fuel lb/min	17.5	15.8	14.1	13.2	12.4	11:9	11.9	11.9
ANM/100 lb	12.8	15:0	17.0	20.0	22.7	26.5	30.0	32.4

# Operating Data - continued

# DIVERSION AND HOLDING DATA

SLOCKS	HEI	HEIGHT	RAN	IGE	(dista	ance	R an	d tin	ne T)	and	ENI	RANGE (distance R and time T) and ENDURANCE (E) for remaining fuel (lb) of	NC	E (E)	for	rem	ainin	g fue	(lb	0 (
R & E         T         E         R         T         E         R         T         E         R         T         E         R         T         E         R         T         E         R         T         E         R         T         T         E         R         T         T         E         R         T         T         E         R         T <th>BLO</th> <th>CKS</th> <th></th> <th>1400</th> <th></th> <th></th> <th>1200</th> <th>12237</th> <th></th> <th>1000</th> <th>•</th> <th></th> <th>800</th> <th></th> <th></th> <th>009</th> <th></th> <th></th> <th>400</th> <th></th>	BLO	CKS		1400			1200	12237		1000	•		800			009			400	
R & E         172         49         63         140         40         51         109         31         40         78         22         29         47         13           Best Ht         30         23         79         236         58         60         170         43         46         98         28         29         47         13           Best Ht         30         20         20         30         10         25         10         10         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M         M         —         M	(II. X	1000)	×	T		×	T	E	×	H	Ξ	R	T .		×	T	Э	×	Н	田
Best R & E         298         73         79         236         58         60         170         43         46         98         28         29         47         13           Best Ht         30         20         10         25         10         10         M		R&E	172	49	63	140	40	51	109	31	40	78	22	29	47	13	17	16	S	9
R & E         236         66         80         194         55         66         153         43         52         111         32         37         69         20           Best R & E         329         77         84         264         63         66         198         43         52         111         32         37         69         20           Best R & E         329         77         84         264         63         66         198         43         52         111         32         37         69         20           Best Ht         30         -         20         30         -         M         30         -         M         20         -         M         M         -         69         20           Best R & E         307         80         92         254         67         75         200         53         59         146         39         43         92         26           Best Ht         30         -         M         30         -         M         30         -         M         -         30         43         92         26           Best Ht         30	SL	Best R & E			4	236	58	09	170	43	46	86	28	29	47	13	17	16	9	9
R & E         236         66         80         194         55         66         153         43         52         111         32         37         69         20           Best R & E         329         77         84         264         63         66         198         43         52         130         34         37         69         20           Best Ht         30          20         30          M         30          M         M          M         M          M         M          M         M          M         M          M         M         M          M <t< td=""><td></td><td>Best Ht</td><td>30</td><td>1</td><td></td><td>30</td><td>1</td><td>10</td><td>25</td><td>1</td><td></td><td>10</td><td>1</td><td>M</td><td>Z</td><td>1.</td><td>M</td><td>Z</td><td>1</td><td>M</td></t<>		Best Ht	30	1		30	1	10	25	1		10	1	M	Z	1.	M	Z	1	M
Best R & E       329       77       84       264       63       66       198       43       52       130       34       37       69       20         Best Ht       30        20       30        M       30        M       M        M       M        M       M        M       M       M        M       <		R&E	236			194		99	153	43		111	32	37	69	1	23	27	00	9
Best Ht         30         -         20         M         30         M         -         M	10	Best R & E		77.	84	264	63	99	198	43	52	130	34	37	69	20	23	27	∞	6
R & E       307       80       92       254       67       75       200       53       59       146       39       43       92       26         Best R & E       354       84       92       289       69       75       225       54       59       154       39       43       92       26         Best Ht       30       M       30       M       30       M       30       M       M       M       M         R & E       378       90       98       312       74       81       246       59       64       180       44       47       114       29		Best Ht	30			_	1	M	30	1	M	20	1	N	Z	1	M	Z	1	Z
Best R & E       354       84       92       289       69       75       225       54       59       154       39       43       92       26         Best Ht       30       M       30       M       30       M       30       M       M       M       M         R & E       378       90       98       312       74       81       246       59       64       180       44       47       114       29		R&E	307						200	53	59	146	39	43	92	26	27	38	11 11	H
Best Ht         30         M         30         M         30         M         30         M <th< td=""><td>20</td><td>Best R &amp; E</td><td></td><td>84</td><td>92</td><td>289</td><td>69</td><td></td><td>225</td><td>54</td><td>59</td><td>154</td><td>39</td><td>43</td><td>92</td><td>26</td><td>27</td><td>38</td><td>=</td><td>=</td></th<>	20	Best R & E		84	92	289	69		225	54	59	154	39	43	92	26	27	38	=	=
R & E 378 90 98 312 74 81 246 59 64 180 44 47 114 29	-11	Best Ht	30		M	30	1	M	30	1	M	25/30	1	X	M	1	M	N	1	Z
	30	R&E	378			312	74	81	246		64	180			114		30	48	13	14

The above table allows for 300 lb fuel in the circuit and includes distance for the descent en route (170 kt., airbrakes out). Enter table at appropriate height block; read off range (R & T) and endurance on the top line. The next line shows the maximum range and endurance at the height shown on the third line (M= maintain height).

If an instrument approach is required, enter the table one column to the right of present fuel. For absolute R & E to flame-out (100 lb allowance for possible gauge errors) enter one column to left. In strong headwinds use of 95% range speeds may be advantageous due to reduced cruise time.

DESCENT DATA

Airbrakes out. Idling or 60% RPM whichever is greater.

Altitude at start of descent (ft)	Time (minutes)	Distance (NM)	Fuel (lb)
35,000	12.2	45	99
30,000	10.8	38	90
25,000	9.3	31	81
20,000	7.7	- 25	70
15,000	6.0	19	58
10,000	4.2	12	44
5000	2.3	6	26

# Abandoning - continued

## **Ejection Drill**

Best speed 250 knots
Convert excess speed into height
Stagger ejections
Head back on rest, elbows in, leave feet on rudder pedals
Close eyes tightly (and if possible lower helmet visor)
Pull seat-pan firing handle
Auto separation occurs below 10,000 feet

#### If the Seat Fails to Fire

Ensure seat-pan firing handle correctly pulled Pull face-screen handle
If seat still fails to fire, bale out manually

#### Manual Bale-Out

Airbrakes IN

Operate CANOPY MDC handle if necessary
Operate manual separation lever
Leave seat and abandon aircraft manually
Pull rip cord D-ring when clear

Reduce speed to practicable minimum

### Failure of Drogue Gun

Above 10,000 feet carry out the manual separation drill Delay pulling rip cord D-ring until at a safe height (no emergency oxygen)

# **Manual Separation**

**MDC** Operation

When forward speed reduced, discard face-screen (if used)
Operate manual separation lever
Fall out of seat
Pull rip cord D-ring

Speed below 350 knots
Canopy closed
Close eyes tightly (and if possible lower helmet visor)
Squeeze CANOPY MDC operating handle and pull firmly
upwards



# Landing Gear Emergency Retraction on the Ground

After Normal Lowering

Raise the guard and operate and hold the EMERGENCY OVERRIDE switch.

Press the UP button.

After Emergency Lowering

Note: Only possible if some hydraulic and electrical power still available.

Return the emergency lowering lever to original setting and carry out the 'After normal lowering' drill.

Barrier Engagement (RAF Mk 5, 6, and 12 and 12A at bight' setting)

Call for barrier
Use steady braking to reduce speed to minimum
HP cock CLOSED
LP cock OFF
STARTING MASTER switch OFF
Aim for centre panel
Keep canopy closed
Duck head forward before engagement
Release brakes before rolling over bottom cable
Resume full braking after engagement
Apply parking brake when stopped
Leave aircraft, avoiding seat firing handles; if practical
make aircraft 'Safe for Parking'.

# Ditching

Do not ditch the aircraft, eject

#### ABANDONING

**Ejection Limitations** 

90 knots/ground level if in straight and level flight

If aircraft is descending, minimum safe height is approximately one-tenth of the rate of descent.

Note: As the first occupant ejects the aircraft may pitch nose-down.

(continued)

Hazardous Landings/ Abandoning

### HAZARDOUS LANDINGS

### Forced Landing

Positioning

Glide at 130 knots (1.5 miles/1000 ft)

Plan for 2500 ft on dead side, abeam touchdown point (High Key). If possible, land on an airfield; if not, consider operating MDC (canopy closed) immediately after touchdown. Do not operate MDC if barrier engagement is likely. Check:

HP cock ... ... CLOSED
LP cock ... ... OFF

STARTING MASTER

switch ... OFF Non-essential electrics Off

Radio ... ... Emergency call
Transponder ... ... Mode A, code 7700
FIRAD checks ... As appropriate
PSP lowering line ... Disconnected
Emergency oxygen tube
Leg restraint cords ... Disconnected

Circuit

Attain 1500 ft downwind opposite touchdown point (Low Key).

Pre-landing checks:

Airbrakes ... IN

Landing gear ... Down on emergency. Three

green lights

Fuel ... ... Cocks off, STARTING MAS-

TER switch OFF. Calculate

threshold speed

Flaps ... ... As required
Harness ... Tight and locked
Wheelbrakes ... Check pressures

Airbrakes may be used to reduce excess speed at roundout.

After Landing

Leave aircraft, avoiding seat firing handles; if practical make aircraft 'Safe for Parking' and apply parking brake.



# Landing Gear Malfunctions — continued

## Landing With Gear in Abnormal Positions

If the Failure to Lower drills prove unsuccessful, no further control over the landing gear is possible. Carry out the following additional pre-landing checks:

Leg restraint cords ... Disconnect PSP lanyard ... Disconnect Emergency oxygen tube Disconnect

Canopy ... ... Leave shut (MDC may be manually operated after touchdown if rapid exit is essential)

Make a normal approach and use the following techniques:

# One Main Leg Unlocked:

Land close to side of runway corresponding to locked down leg. At touchdown:

HP cock ... ... CLOSED
LP cock ... ... OFF
STARTING MASTER
switch ... ... OFF

After touchdown, hold wings level by use of aileron. When wing drops, counteract the resulting swing with brake

# Nose Leg Unlocked

At touchdown:

HP cock ... ... CLOSED LP cock ... OFF

STARTING MASTER

Lower nose onto runway before longitudinal control is lost. Use brake to keep straight.

OFF

Note: Do not engage barrier if MDC has been operated or with canopy open.

# Wheels-Up Landing

Land on a foam carpet if possible. Make a normal approach and fly the aircraft onto the runway at the normal speed.

Hydraulic/ Landing Gear

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At touchdown:

HP cock ... ... CLOSED LP cock ... ... OFF STARTING MASTER

switch ... ... OFF Fire extinguisher ... Operate

#### HYDRAULIC FAILURE

#### Indication

SWS operates, HYD caption on

#### Actions

Cancel SWS

Check brakes pressure gauge

Retract airbrakes if extended: do not use again

Lower flap fully when convenient, but before lowering the

landing gear

If no flap, increase normal threshold speed by 10 knots Lower the landing gear on the emergency system.

During the landing run use brakes in one continuous application.

### LANDING GEAR MALFUNCTIONS

# Failure to Lower-Electrical and/or Mechanical Failure

Indications

Landing gear indicators out or red

Operate indicator light changeover switch

Check hydraulic pressure in green sector. HYD caption

Re-cycle the landing gear

Apply positive and negative g

Yaw the aircraft

Considerations =

If hydraulic pressure is insufficient or the above actions are not successful, select DOWN on the normal landing gear system then lower the landing gear using the emergency system.

# Self Cycling

Action

Select the landing gear down and operate the emergency lowering lever.

# COMMUNICATIONS FAILURE

Complete Loss of Sound (no side tone)

Action

Check helmet mic-tel connection

On CCU select:

NORM/FAIL switch to FAIL

NORM/EMERG switch to EMERG

Rotary switch to COMM

COMM toggle switch up, all other toggle switches down

Confirm intercom available

If no intercom select:

STANDBY UHF POWER switch to NORMAL

CCU rotary switch to UHF SBY

UHF SBY toggle switch up, all other toggle switches

Confirm intercom available

If no intercom select:

STANDBY UHF POWER switch to EMERG

# Loss of Transmit or Reception Facility

Action

Check mute switch and transmit buttons

Select CCU to emergency as above

Try alternative aerial

Dial frequency manually

Change frequency

Select TR+G

Dial 243 MHz

Select CCU rotary switch to UHF SBY, COMM toggle switch down, UHF SBY toggle switch up.

If standby radio fails, set UHF POWER switch to EMERG

If complete UHF failure occurs, set the transponder to Mode A and code selectors to 7600

If standby radio selected, ie main radio inoperative, select CCU NORM/FAIL switch to NORM and NORM/EMERG switch to NORM in order to regain full CCU facilities

UHF Mk 5/ Comms Mk 5A

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# TMk 5

### UHF FAILURE

Complete Loss of Sound (no side tone)

Action

Check mic-tel connections

Select UHF NORMAL/STANDBY switch to STAND-BY (20 sec warm-up period)

Check electrics

## Loss of Transmit or Receive Facility

Action

Check mute switch and buttons

Select other aerial

Try alternative transmit switch

Dial frequency manually

Change frequency

Select 243 MHz and T/R+G together

Switch off

Select UHF NORMAL/STANDBY switch to STANDBY (20 sec warm-up period)

If power to standby set fails, select STANDBY UHF NORMAL/EMERG switch to EMERG

If complete UHF failure occurs set the transponder to Mode A and the code selectors to 7600.

# SPEECHLESS PROCEDURE

YES

NO

· · SAY AGAIN

HOMING

- FURTHER EMERGENCY

CARRIED OUT

FIELD IN SIGHT



Card 18 (AL1)

Oxygen Failures — continued

Difficulty in Breathing Out

Lift mask off face

Press TEST MASK button to clear inhalation valve If unsuccessful, breathe in through mask and exhale with mask away from face.

Descend below 10,000 ft cockpit altitude

Continuous white MI

Check mask fit
Check tube for leaks
Check contents

If not cured, carry out Emergency Oxygen Drill
Continuous black MI

Pressure over/under 200 to 400 PSI Contents in red sector SWS operates, OXY caption on Carry out Emergency Oxygen Drill

AIR CONDITIONING AND PRESSURISATION

Under-pressurisation or Canopy Failure

Indications

Above 26,000 ft cockpit altitude, SWS operates, CPR caption on

Actions

Cancel SWS

Select 100% oxygen

Descend immediately below 35,000 ft cockpit altitude to prevent hypoxia and then to below 25,000 ft cockpit altitude to reduce risk of decompression sickness. Check for air flow from open punkah louvres.

No air flow indicates failure of air conditioning system, therefore, no cockpit temperature or windscreen demisting facilities.

Over-pressurisation

Indications

Cockpit altitude less than  $\frac{1}{2}$  actual altitude + 2000 ft

Actions

Select UNPRESS

Proceed as for Under-pressurisation or Canopy Failure

**Temperature Control Failure** 

If cockpit conditions become uncomfortable, select pressurisation off.

Proceed as for Under-pressurisation or Canopy Failure
If ventilation is required, select EMERG DEPRESS

Oxygen/ Air cond.

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Electrical

(contd.)/

#### Electrical Failures — continued

Action

Check flight instruments START/OFF switch at START Operate flight instruments switch to STANDBY; if standby inverter does not start, return switch to NORMAL and attempt to re-start normal inverter by operating inverter test switch.

If inverters remain off-line, return test switch to OFF.

Considerations

The following services will be lost:

Mk 22 altimeter Top temperature control Artificial horizon Altimeters vibrators

Gyro compass

Fire detection

Cabin temperature control system
Instrument panel lighting (Mk 5 only)

### Suspect Inverter

Indications

Degradation of aircraft flight instruments performance. Inverter MI alternating black/white.

Action

Operate FLIGHT INSTRUMENT-NORMAL/STAND-BY switch to STANDBY.

If practicable descend below cloud.

# **♦ Lighting Failure** (Mk 5A only)

(a) Lighting inverter failure

Action

Select EMERG LTS switch to MAIN BATT.

(b) Power failure (main battery failure)

Action

Select EMERG LTS switch to EMERG BATT position by pulling switch out and down.

#### OXYGEN FAILURES

# Emergency Oxygen Drill

Operate emergency oxygen bottle
Disconnect main supply
Descend below 10,000 feet cockpit altitude
Note: Breathing out will be difficult.

# Hypoxia

If hypoxia suspected, carry out Emergency Oxygen Drill

# Difficulty in Breathing In

Check connections, tube for kinks
If not cured, carry out Emergency Oxygen Drill

continued



### **ELECTRICAL FAILURES**

#### Generator Failure

Indications
SWS operates, GEN caption on
Inverter MI white

Actions
Cancel SWS
Non-essential electrics off
STANDBY UHF

POWER switch ... NORMAL Mk 5 UHF selector ... STANDBY

- or -

Mk 5A (rotary switch ... UHF SBY selected CCU ) toggle switches UHF SBY selected

Main UHF ... OFF when standby set warmed up

Keep UHF transmissions to a minimum Land as soon as practicable Before the main batteries fail, set:

Mk 5 T & S SUPPLY STANDBY

Fuel/ Electrical

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Mk 5A ATT IND STANDBY/NORMAL switch ... STANDBY

STANDBY UHF

POWER ... EMERG

Oxygen ... 100%

If the emergency battery is in use, lower the landing gear on the emergency system

### Normal Inverter Failure

Indication

Inverter MI white (No GEN warning on SWP)

Actions

Check altimeter and artificial horizon OFF flags retracted. Operate flight instruments switch to STANDBY and back to NORMAL.

If MI remains white, no further action is possible If practicable, descend below cloud.

### **Double Inverter Failure**

Indications
Inverter MI white
Altimeter and artificial horizon OFF flags showing

### **FUEL SYSTEM FAILURES**

### **Fuel Tank Pressurisation Failure**

Indication

Fuel tank air pressure less than 1.5 PSI in level flight.

Action

Avoid negative g. Land as soon as practicable.

#### Considerations

- 1. Maximum RPM possible without fluctuation decreases as fuel is used and altitude is reduced.
- 2. Below 5000 feet, with zero fuel tank pressure and 250 lb fuel, maximum RPM obtainable is approx 90%.
- 3. Fuel gauge may overread.
- 4. Fuel in tip tanks will not transfer. ▶ ◀

## **Excessive Fuel Tank Air Pressure**

Indication

Fuel tank air pressure above 2.5 PSI.

Action

Reduce speed and RPM
Descend and land as soon as practicable
Avoid increased g loadings

# **Booster Pump Failure**

Indications

SWS operates, FUEL caption on

Action

Cancel SWS

Post-mod 1772, check starter master switch ON

◆ If warning remains on select starter master switch OFF. ▶

Avoid negative g and land as soon as possible

### Considerations

- 1. Maximum RPM may not be available above 12,000 feet.
- 2. Above 12,000 feet at high throttle settings and in high ambient temperatures, RPM may fluctuate and a flame-out is possible.
- 3. Above 30,000 feet an immediate flame-out is possible.
- 4. A successful relight above 10,000 feet (AVTUR) or 6000 feet (AVTAG) is unlikely.

#### FLAME-OUT

Immediate Actions (Immediate relight)

Throttle ... ... CLOSED

Press relight button for a maximum of 10 seconds. If after

10 seconds no relight has occurred, set:

Non-essential electrics ... Off

Transponder and RT ... As required Glide at 130 knots (1.5 NM/1000 feet)

Cold Relight (Maximum recommended height 15,000 feet)

Note 1: If the booster pump has failed, a successful relight is unlikely above 10,000 feet (Aytur): 6000 feet (Aytag).

Note 2: Mechanical noise from the windmilling engine is noticeable. At high airspeeds, intake banging may occur; the intensity of the banging decreases with reduced airspeed and height but it may still be present at 120 knots.

Check:

Battery switch ... ON

STARTING MASTER

switch ... ... ON
Ignition switch ... ... ON

Rain/ice clearance ... OFF

Pressurisation ... Select UNPRESS

LP cock ... ... ON (FUEL caption out)

HP cock ... ... CLOSED
Throttle ... ... CLOSED

Set speed 120 to 170 knots.

Press the relight button and at the same time open the HP cock. Keep the button pressed for up to 30 seconds or until RPM reach 40%. If flame-out was caused by icing, make careful throttle movements and maintain 70% RPM for at least 5 minutes.

# Failure to Relight

If, after 30 seconds no relight has occurred, release the relight button and close the HP cock. If possible, allow one minute to elapse before attempting a further relight. There may be fumes in the cockpit.

Eng Fail (contd)/ ng Icing/ Flame-Out

### ENGINE FAILURE AFTER TAKE-OFF

### Below 160 Knots or 600 feet AGL

Land within 180° are ahead or, if conditions within limits, eject.

If aircraft descending, minimum safe height for ejection is approximately one-tenth rate of descent.

### Above 160 Knots or 600 feet AGL

Make level turn 40° to 50° bank (into crosswind) towards airfield. Check:

HP cock ... ... CLOSED
LP cock ... ... OFF

STARTING MASTER

switch ... ... OFF

Radio ... Emergency call

Glide at 130 knots. Lower landing gear on emergency system and use flaps as required.

Do not operate MDC if barrier engagement likely. Leave aircraft, avoiding seat firing handles: if practical make aircraft 'Safe for Parking' and apply parking brake.

### ENGINE ICING

### Indications

Any, or all of the following indications may occur:

Loss of power Increase in JPT Reduction in max obtainable RPM Engine surge Flame-out

#### Actions

If it is not possible to leave the icing conditions, all throttle movements must be made carefully and a maximum of 85% RPM used.

If the engine surges, and height permits, throttle back and descend at maximum rate to below freezing level. Set 70% RPM for 5 minutes, then increase RPM carefully. An abnormally rapid rise in JPT indicates incomplete de-icing; set 70% RPM for a further period.

If flame-out occurs, do not normally attempt to relight until clear of icing conditions. After relight, de-ice the engine as above. Avoid rapid throttie movements. Card 15 (AL4)

#### ENGINE FIRE ON THE GROUND

Throttle ... ... Closed
HP cock ... ... CLOSED
LP cock ... ... OFF

STARTING MASTER

switch ... OFF Extinguisher ... Operate

Release parking brake and vacate aircraft quickly. When fire is extinguished, and if possible replace seat and canopy pins.

Fire/ Fumes/ Eng Failure

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#### FIRE IN THE COCKPIT

Carry out Noxious Fumes in Cockpit drill below. If fire/smoke has occured shortly after an electrical switch has been moved, reselect the particular switch to its former position.

If fire/smoke persists: All non-essential electrics off. If visible signs of fire, use cockpit fire extinguisher.

### NOXIOUS FUMES IN COCKPIT

Select 100% oxygen Regulator toggle to EMERGENCY Select EMERG DEPRESS Pressurisation OFF Descend below 25,000 feet cockpit altitude Check for signs of fire If required:

14

With canopy closed, operate MDC below 350 knots: do not engage barrier if MDC has been operated or with canopy open.

## ENGINE MECHANICAL FAILURE

Throttle ... ... CLOSED
HP cock ... ... CLOSED
LP cock ... ... OFF

STARTING MASTER

switch ... ... OFF Non-essential electrics ... Off

Carry out a forced landing or eject

Do not attempt to relight

Engine Failure After Take-Off - overleaf

### ENGINE FIRE

### Indications

SWS and audio operates FIRE caption on

#### Immediate Actions

Throttle CLOSED (if impracticable,

eject) SWS Cancel

Reduce to practicable mini-Speed

Check for confirmatory signs of fire

### Subsequent Actions

1. If the FWL goes out within 5 seconds and there are no other signs of fire (hot gas leak):

Land as soon as possible at suitable airfield, using minimum power. If warning recurs, throttle back at frequent intervals to check that warning goes out.

2. If there are definite signs of fire:

HP cock ... CLOSED LP cock ... OFF

**4** STARTING MASTER

switch ... OFF

Operate (below 200 kt for Extinguisher best results)

Noxious fumes drill Radio call and Transponder to EMERG

As situation dictates

If fire goes out ... FWL out within 30 seconds If warning remains on ... Eject

3. If the FWL stays on but there are no other signs of fire:

On the evidence available, a decision must be made whether to treat the warning as real or spurious.

Decision Action Real Complete fire drill (see 2 above)

Spurious Land as soon as possible at suitable airfield using minimum power. Be prepared to resume fire drill or eject

if further signs of fire

Note 1: Do not test FWL when airborne.

Note 2: The attention-getter will be re-activated if the warnings are still on when inverter changeover occurs.

Note 3: Operation of the fire extinguisher causes a muffled bang and possibly a pungent smell.

Note 4: Do not re-start engine.

Note 5: Post-mod 1772, booster pump is controlled by START-ING MASTER switch.

B-2305-14

Card To (AL1)

# JET PROVOST TMk 5 & 5A

# **EMERGENCIES**

ENGINE FIRE
NOXIOUS FUMES
ENGINE MECHANICAL FAILURE
ENGINE FAILURE AFTER TAKE-OFF

ENGINE ICING FLAME-OUT

FUEL SYSTEM FAILURES ELECTRICAL FAILURES

ELECTRICAL FAILURES (contd)
OXYGEN FAILURES
AIR CONDITIONING & PRESSURISATION

T Mk 5—UHF FAILURE
T Mk 5A—COMMUNICATIONS FAILURE

HYDRAULIC FAILURE
LANDING GEAR MALFUNCTIONS

HAZARDOUS LANDINGS ABANDONING