APO YARKER Card 1 October, 1973 (AL3, Sept 75) AP TOTB-3801-14 4: Homing (Issue 1) Speedless procedure: 3. Say ayrin x: Further emergency Fuel attymmetry net to excess 3 ink galling for sprinning FLIGHT REFERENCE CARDS F.P. 62/3" at 1000; notice by 3" per 1000' 12 man; 91/2 probably blocked what needle. BULLDOG T Mk 1 NORMAL DRILLS Temps during flight: O.P 60-90 psi 0.7 60 -118°C CH.T. SO - 246°C Prepared by Procurement Executive, Ministry of Defence, in collaboration with RAF Handling Squadron Memelor failure: ill F -12 min VHF -35 min, BY COMMAND OF THE DEFENCE COUNCIL Ledwed arispered: Kon was 2200 lpm win wix. Turbulence: 85 Kt Broster pump on Bad Viss

#### NOTES TO USERS

- 1 These Flight Reference Cards are complementary to the Bulldog T Mk 1 Aircrew Manual (AP 101B-3801-15). The same conventions are used and amendment procedure is similar.
- 2 Comments and suggestions regarding these Flight Reference Cards should be forwarded to the Officer Commanding, RAF Handling Squadron, Boscombe Down, Salisbury, SP4 0JF.

# LIST OF CARDS

| Card                                 | Issued   | Card                                  | Issued   |
|--------------------------------------|--|---------------------------------------|--|
| No                                   | by   | No                                    | by   |
| 1<br>2<br>3<br>4<br>5<br>6<br>7<br>8 | AL3<br>AL3<br>AL3<br>AL3<br>AL3<br>AL3<br>AL3<br>AL3 | 9<br>10<br>11<br>12<br>13<br>14<br>15 | AL1<br>Initial Issue<br>AL3<br>AL3<br>AL3<br>AL3<br>AL3<br>AL3 |

#### INITIAL CHECKS

Initial checks On approaching the aircraft check:

General position ... Clear of other aircraft

No fuel or oil leaks

Ground fire extinguisher Available

Chocks ... In position

Before commencing the external checks carry out the

following checks in the cockpit:

Ignition ... ... OFF Battery master switch ... OFF

Throttle ... Closed

Mixture ... CUT-OFF

Flying control locks ... Removed and stowed

**♦** Starter circuit breaker ... Tripped (aircraft without

starter button cover)

# EXTERNAL CHECKS

Carry out a systematic check of the aircraft for obvious signs of damage, leaks, loose panels or fairings. In particular, check:

Canopy ... ... Condition and operation Cockpit transparencies ... Condition

# Left landing gear

Fairings ... Condition, secure

Brake lead ... Secure

Tyre ... Examine for cuts, creep and

pressure



External

checks

# External checks - continued

# Left mainplane

Flap ... Up. Linkages secure

Aileron ... Full and free movement, link-

age secure

Navigation light ... Condition of cover

Pressure head ... Cover removed, free of ob-

struction

Fuel tank filler cap ... Access panel flush and secure

# Engine

Cowling ... ... Fasteners secure

Intakes and ducts ... Clear

Spinner ... ... Condition, secure

Propeller ... Condition Exhaust pipes ... Secure

# Nose landing gear

Nosewheel straight

Steering torque link ... Connected

Shock absorber ... Correct extension

Tyre ... Examine for cuts, creep and

pressure

# Right mainplane

As for left mainplane except pressure head Landing and taxy lamps Condition

# Right landing gear

As for left landing gear

#### Tail unit

Elevator ... Full and free movement, link-

age secure

Rudder ... ... Secure, trim tab secure, link-

age secure

Do not move the rudder by hand

Tail bumper ... Undamaged

# COCKPIT CHECKS

On entering the cockpit, check:

Seats ... ... Backs adjusted and locked Harnesses ... Condition and security

Right harness connected and tightened if flying solo

**♦** Canopy jettison handle Indicators aligned

Hand fire extinguisher, first aid kit, axe and asbestos gloves ...

Stowed and secure Strap secure

Baggage (if carried) ... Secure

Loose articles ... None

Adjust and lock the rudder pedals evenly

continued

# Cockpit checks - continued

Strap in Check or select:

Propeller ... ... Clear Battery master switch ... ON

External power ... On (if required)

CCU ... On, volumes adjusted

Receiver and transmitter switches as required NORM/FAIL switch NORM

NORM/EMERGY switch

NORM

Internal and external

lighting ... As required

Parking brake ... On

Circuit breakers ... All made (in)

◆ Pressure head heater ... OFF

UHF aerial switch ... As required

Accelerometer ... Reset to +1.0 g Clock ... Working, correct

Flap indicator ... Correct indication Volt/amp selector ... V, check voltage

Fuel contents ... Sufficient and correct

Magnetic compass ... Condition Flight instruments ... Condition

Altimeter set to zero

Engine instruments ... Condition Vacuum gauge ... Condition

VHF and UHF ... ... OFF

Cabin heat controls ... As required

Induction hot air ... COLD

Fuel booster pump ... OFF

Alternator ... ON (OFF if external power in

use)

Alternator warning light On

Starter warning light ... Press to test, then out

Throttle / RPM lock lever Fully forward Exercise, set closed

RPM control ... Exercise, set maximum

Mixture ... Fully rich

Elevator trim ... Full and free movement

Set in TO band

Rudder trim ... Full and free movement

Fuel selector valve ... LEFT

Flying controls ... Ailerons and elevator — full, free and correct movement

Cockpit checks

#### STARTING THE ENGINE

# Cold engine

To start the engine from cold, set the anti-collision light on and give the start-up signal to ground crew. When clear, select or check:

Ignition ... ... OFF

\* Starter button ... Press, count 8 blades of propeller, then release

Throttle ... ... \frac{1}{4} inch open

Fuel booster pump ... ON, then OFF (see Note)

Ignition ... ... L

Starter button ... Press until engine starts

Note: It is easy to overprime the engine. The fuel booster pump should not be ON for more than 1 to 2 seconds.

\* For aircraft's first flight of day only.

# When the engine starts:

Starter button ... Release Starter warning light ... Out Ignition ... BOTH

# Hot engine

To start the engine from hot, set:

Anti-collision light ... On Mixture ... CUT-OFF

Throttle ... 1 inch open (see Note)

Note: When CHT is 50°C or above, up to half throttle may be required.

Give start-up signal to groundcrew. When clear, select ignition to L (left) and push the starter button fully in until the engine starts

# When the engine starts:

Starter button ... Release
Starter warning light ... Out
Mixture ... Fully rich
Throttle ... Closed
Ignition ... BOTH

If the starter warning light does not go out the engine should be closed down and the fault investigated

#### FAILURE TO START

If the engine fails to start after 10 to 12 seconds, release the starter button. Check the fuel booster pump is OFF, then wait for 5 minutes before making a further attempt to start. If the cause of the failure to start is overpriming, make the next attempt to start as for starting a hot engine.

# AFTER STARTING

Oil pressure ... 25 PSI within 30 seconds

External power ... Disconnected

Alternator ... On

#### Set 1200 RPM

Alternator warning light Out Voltage ... ... 28 V

Ammeter ... Positive reading

Pressure head heater ... ON, of groundcrew check, then OFF

Volt/amp selector ... Yhups Vacuum gauge ... Indicating

Horizon and DI ... Erecting, DI aligned with

VHF and UHF ... magnetic compass
On, frequencies selected

Ignition ... ... Check for live and dead magneto

Flap ... ... Correct operation

Fuel selector valve ... RIGHT VHF and UHF ... Test Altimeter ... Set QFE Canopy ... Latched

**Engine** starting

# TESTING THE ENGINE

| ALDINIO  | IIII EITOITE   |
|--|--|
| Aircraft into wind   |  |
| Parking brake  | On   |
| Control column   | Central  |
| Fuel selector valve  | BOTH   |
| Oil temperature  | 30°C min   |
| Oil pressure   | 25 PSI min   |
| Cylinder head temp   | 100°C min  |
| Set 1800 RPM   |  |
| RPM control (three times   |  |
| for first flight of day)   | Move towards minimum until                           |
|  | RPM decrease   |
|  | Note: Do not allow RPM to de-                        |
|  | crease by more than 500 RPM.                         |
|  | Return to max  |
| G . 4400 P.P. 5  | Check RPM restored                                   |
| Set 2100 RPM   | CI 1 1 175 DDV                                       |
| Magnetos   | Check max drop 175 RPM                               |
|  | Max drop diff between mags                           |
|  | 50 RPM   |
|  | Check RPM restored at BOTH                           |
| Induction hot air  | Set HOT — RPM decrease                               |
|  | Set COLD — RPM restored                              |
| Vacuum   | 4½ to 5 in Hg  |
| Throttle   | Close—idling RPM 720 to 730                          |
|  | 750 ± 25   |
| Trims: TAKE-O  | FF CHECKS  |
| Elevator   | In TO band   |
| Dudden   | TO   |
| Throttle friction  | Adjusted   |
| RPM control  | Maximum  |
| 7.7.   | Fully rich   |
| To deside a beat also  | COLD   |
| Fuel:  | COLD   |
| Booster pump   | ON   |
| Combont  | Sufficient   |
| C-1  | BOTH   |
| The state of the s | INTER  |
| Flap Instruments:  | INTER  |
|  | As required  |
| Taxy lamps Pressure head heater  | ON D   |
| T.7.   | Indicating   |
|  | Erect, DI synchronised                               |
| Flight instruments   |  |
| Engine instruments   | Temperatures and pressures                           |
| Alternatorwarninglight   | Out  |
| Harness  | Tight  |
| Canopy   | Latched  |
| Flying controls  | Aileron and elevator full and                        |
| Note: The gudder should b  | free movement<br>have been checked for full and free |
| movement during taxying.   | nave been enceased for full and free                 |
| motoment during majing.  |  |

Card 6 (AL3)

#### AFTER TAKE-OFF

Engine instruments ... Checked

Flap ... ... UP at safe height

\*Fuel booster pump ... OFF at transition altitude

\* When clearing the circuit.

# CHECKS BEFORE STALLING, SPINNING, AEROBATICS

# Height

Sufficient to recover by briefed height

#### Airframe

Flap ... ... UP for spins and aerobatics Elevator trim ... In take-off band for spins

# Security

Canopy ... Closed and locked

Harness ... Tight Loose articles ... None

#### Engine

RPM control ... 2600 RPM
Mixture ... Best power
Induction air ... COLD
Fuel contents ... Satisfactory
Fuel selector valve ... BOTH

Instruments ... Indications normal

Engine test/ Take-off/ In flight

#### Location

Clear of controlled airspace, active airfields, built-up areas and cloud

#### Lookout

Clear of other aircraft

# Recommended speeds for aerobatics

 Roll
 ...
 ...
 130 kt

 Barrel Roll
 ...
 120 kt

 Stall Turn
 ...
 120 kt

 Loop
 ...
 140 kt

 Half Roll off Loop
 ...
 145 kt

# CHECKS AFTER STALLING, SPINNING, AEROBATICS

DI ... Synchronised

#### APPROACH PROCEDURE

|                 | Configuration | RPM control | Throttle set to | IAS<br>kt |
|-----------------|---------------|-------------|-----------------|-----------|
| Initial descent | Flap UP       | 2400        | 2200 RPM        | 100       |
| Glidepath       | Flap INTER    | Max         | As reqd         | 80        |

When in visual contact with the runway, set FULL flap and reduce to threshold speed + 5 kt. Fly the GCA pattern at 100 kt, flap UP.

# Checks before joining the circuit

| Fuel    |       | <br>Contents sufficient         |
|---------|-------|---------------------------------|
|         |       | ◆ Booster pump on below transi- |
|         |       | tion level                      |
| Instrun | nents | <br>Erect and synchronised      |

Radio ... ... Correct frequency

Altimeter ... QFE set

# LANDING CHECKS

# Checks before landing

+ec+ 0H

| RPM control .     |   | Max        |
|-------------------|---|------------|
| Mixture .         |   | Fully rich |
| Induction hot air | r | COLD       |

Fuel ... ... Booster pump ON Contents sufficient

Flap ... ... As required
Harness ... Tight
Canopy ... ... Latched
Brakes ... Off

Threshold speeds are given on Card 7

continued

# Landing checks - continued

# Threshold speeds

|               | FULL flap | Flap UP |
|---------------|-----------|---------|
| Powered       | 65        | 70      |
| Glide         | 75        | 75      |
| Short landing | 55        | _       |

These speeds are for an AUW of 2350 lb and may be reduced by 1 kt per 100 lb reduction in AUW.

For crosswind and/or turbulent conditions, add 5 knots

# Checks after landing (aircraft stationary)

14

Pressure head heater ... OFF Fuel booster pump ... OFF

**√** Taxy lamps ... As required

Flaps ... ... UP

# SHUTDOWN

Parking brake ... On

Set 1000 RPM and allow temperatures and pressures to stabilise. When cylinder head temperature is below 180°C:

Throttle ... ... Closed Mixture ... CUT-OFF

Ignition ... ... OFF (when engine stops)

Fuel selector valve ... LEFT or RIGHT

VHF and UHF ... Off
Alternator ... OFF
Electrical services ... Off

Intercom ... ... Off
Battery master switch ... OFF
Chocks ... ... In position

Parking brake ... Off

Approach/ Landing/ Shutdown

# AIRFRAME LIMITATIONS

| Maximum speeds   |                                 |                |                  |         |  |
|--|---------------------------------|----------------|------------------|---------|--|
| Never exceed speed<br>Normal operating spe<br>For full application of<br>Flap: between UP ar<br>between INTEL<br>Canopy open (max 8) | ed<br>aileron<br>d INT<br>R and | ER             | or or r<br>      | udder   | 185 kt<br>145 kt<br>140 kt<br>135 kt<br>100 kt<br>120 kt |
| Max OAI (sea level)  |                                 |                |                  |         | 0,000 ft<br>+35°C  |
| Min OAT (sea level) Aerobatics and spinning  |                                 | <br>ibited v   | when:            | •••     | −10°C  |
| Flaps are extended<br>CG is aft of 68.9 in AC  |                                 |                | ggage<br>PSP is  |         |  |
| Weight   |                                 |                |                  |         |  |
| Max for take-off and Max zero fuel weight (card 8) Note: If ZFW is above B and C. Max load in baggage                                | re 2050                         | lb, opera      | ation is         | rea A   | 2350 lb<br>2050 lb<br>1 to areas<br>220 lb               |
| Airfield limitations   |                                 |                |                  |         |  |
| Max crosswind compo<br>Aircraft approach limi<br>PAR<br>Runway aids with<br>glidepath guidar   | itations<br>20<br>out           | :<br>00 ft tru | e or in          | ndicate | d AGL  |
| Arrester gear tramplin   | ng:                             |                |                  |         |  |
| SPRAG Erected  | up to                           | 30 kt, n       | o brak<br>i be c | ing ap  | at right<br>plied<br>only at                             |
|  |                                 |                |                  |         |  |

# Airframe limitations (contd.)

#### CG and Normal-acceleration

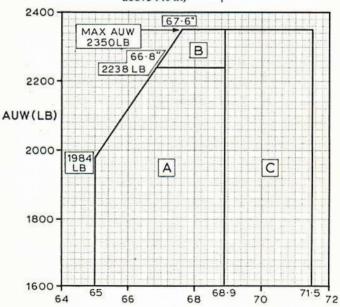
The CG must always lie within the outline boundaries of the envelope shown below.

The normal max permitted positive and negative normalaccelerations are a function of CG position, weight and speed as given below.

Note: Never exceed limits are given in the Aircrew Manual. Clean aircraft

# 1. Symmetric manoeuvres

|          |   | Aerobatics permitted                  | Non-aerobatic                       |   |
|----------|---|---------------------------------------|-------------------------------------|---|
| Normal   | A | +5.25 to -2.25 (-1.0<br>above 140 kt) | <b>4C</b> +3·0 to−0·75 (+0·25 above |   |
| g limits | В | +3.75 to -1.0 (-0.5<br>above 140 kt)  | 140 kt)                             | P |



# CG POSITION INCHES AFT OF DATUM

 Rolling manoeuvres using full aileron at speeds up to 140 kt

Normal A +3.5 to +0.5
B +2.5 to +0.5

◆C Limited to normal turns using low rates of roll ▶

Flaps extended at any setting

Max normal-acceleration ... +2.0g

Limitations



# **ENGINE LIMITATIONS**

#### **RPM**

| Max | permissible |       |       | ••• | ••• | 2700 |
|-----|-------------|-------|-------|-----|-----|------|
| Max | (oil temp b | pelow | 30°c) |     |     | 1200 |

# Maximum manifold pressure (in Hg)

| RPM                         | 1800 | 1900 | 2000 | 2100 | 2200 | 2300 |
|-----------------------------|------|------|------|------|------|------|
| Sea level                   | 25.0 | 25.6 | 26.3 | 26.9 | 27.5 | 28.2 |
| Reducing to                 | 23.2 | 24.1 | 25.0 | 26.0 | 26.9 | 27.8 |
| At (altitude)<br>(See Note) | 6000 | 5000 | 4000 | 3000 | 2000 | 1000 |

Note: These altitudes correspond to full throttle height for the associated RPM

# Magneto drop at 2100 RPM

May each magneto

| Max each magneto   | 1/3 |
|--|-----|
| Max between magnetos   | 50  |
| Cylinder head temperature (°C)                                   |     |
| Max at full throttle   | 246 |
| Max before shutdown (after flight)                               | 180 |
| Min before exceeding 1200 RPM (following engine start on ground) | 100 |
| Min during flight  | 50  |
| Oil pressure (PSI)   |     |
| Min at idling RPM  | 25  |
| Min (normal operation)   | 60  |
| Max (normal operation)   | 90  |
| Max during start and warm up                                     | 100 |
| Oil temperature (°C)   |     |
| Maximum  | 118 |
| Min (continuous operation)                                       | 60  |
|  |     |

30

12

Min before exceeding 1200 RPM

Fuel pressure (PSI)

Maximum

Limitations (contd)

# CRUISE PERFORMANCE—BEST POWER MIXTURE

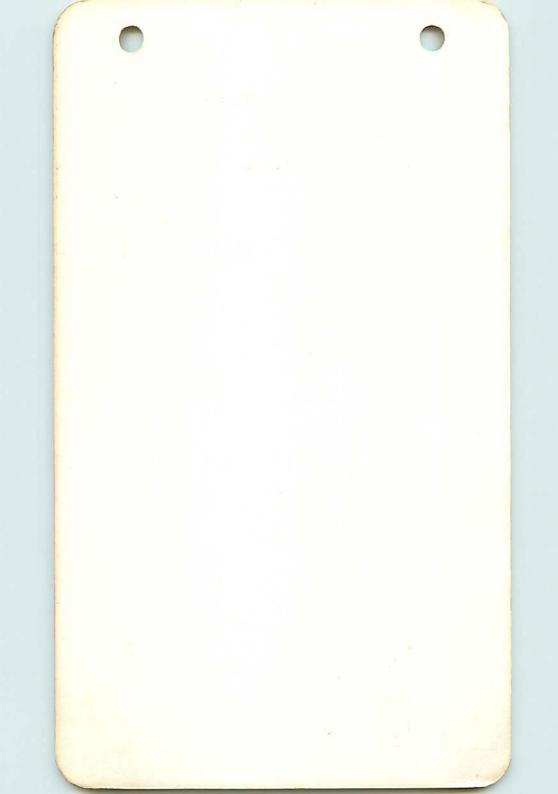
| Pwr % |             |      | IAS  | (Kt)        |                |      | Coi      | Fuel<br>rsump | tion |
|-------|-------------|------|------|-------------|----------------|------|----------|---------------|------|
|       | 2300 lb auw |      |      | 1900 lb auw |                |      | (Gal/hr) |               |      |
|       | SL          | 4000 | 8000 | SL          | 4000           | 8000 | 2600     | 2400          | 2200 |
| 95    | 134         | _    | -    | 135         | -              | _    | 12.5     | -             | 2-0  |
| 90    | 130         | _    | -    | 132         | <del>2</del> 2 | _    | 11.8     | -             | 1    |
| 80    | 123         | 120  | _    | 126         | 123            | _    | 10.7     | 10.5          | _    |
| 70    | 116         | 113  | 110  | 120         | 116            | 113  | 9.7      | 9.4           | 9.1  |
| 60    | 108         | 107  | 103  | 113         | 111            | 107  | 8.7      | 8.4           | 8-1  |
| 50    | 100         | 97   | 95   | 104         | 102            | 101  | 7.7      | 7.4           | 7.1  |
| 40    | 89          | 86   | 83   | 94          | 91             | 89   | 6.7      | 6.4           | 6.1  |

# CRUISE POWER SETTINGS — BEST POWER MIXTURE

|     |       | 2600 | 2600 RPM  |      |       | 2400 RPM | RPM       |      |       | 2200 RPM | RPM       |      |
|-----|-------|------|-----------|------|-------|----------|-----------|------|-------|----------|-----------|------|
| Pwr | ŗ     |      | Man Press | SS   | 2     | W        | Man Press | Was- | 22    | W        | Man Press | SS   |
| e - | Press | SL   | 4000      | 8000 | Press | SI       | 4000      | 8000 | Press | SF       | 4000      | 8000 |
| 95  | 6.4   | 28.5 | 1         | 1    | I     | 1        | 1         | 1    | 1     | - 1      | 1         | 1    |
| 90  | 5.7   | 27.5 | 1         | l    | 1     | f        |           | 1    | 1     |          | -         | I    |
| 80  | 4.7   | 25.0 | 24.0      | 1    | 4.5   | 27.0     | 1         | 1    | 1     | 1        | 1         | 1    |
| 02  | 3.9   | 22.5 | 21.5      | 20.5 | 3.7   | 24.5     | 23.5      | 1    | 3.5   | 26.5     | 25.5      |      |
| 09  | 3.1   | 20.0 | 19.5      | 18.5 | 2.9   | 21.5     | 21.0      | 20.0 | 2.7   | 23.5     | 22.5      | 21.5 |
| 20  | 2.5   | 17.5 | 17.0      | 16.0 | 2.3   | 19.0     | 18.0      | 17.5 | 2.1   | 20.5     | 19.5      | 19.0 |
| 40  | 1.9   | 15.5 | 14.5      | 14.0 | 1.7   | 16.5     | 15.5      | 15.0 | 1.6   | 17.5     | 17.0      | 16.0 |

# RANGE - BEST POWER MIXTURE

| Fuel  | Pwr<br>% | Range — NM |      |      |          |      |                |          |      |      |
|-------|----------|------------|------|------|----------|------|----------------|----------|------|------|
|       |          | 2600 грм   |      |      | 2400 грм |      |                | 2200 крм |      |      |
| (Gal) |          | SL         | 4000 | 8000 | SL       | 4000 | 8000           | SL       | 4000 | 8000 |
|       | 95       | 247        | =    | -    | _        | _    | _              | _        |      | _    |
| 28    | 90       | 256        | _    | -    | _        | _    | _              | _        | _    | _    |
|       | 80       | 268        | 278  | -    | 274      | -    | -              | -        | -    | _    |
|       | 70       | 283        | 293  | 302  | 290      | 302  | ( <del>-</del> | 300      | 312  | -    |
|       | 60       | 298        | 309  | 319  | 307      | 319  | 331            | 319      | 331  | 343  |
|       | 50       | 312        | 321  | 334  | 324      | 334  | 348            | 338      | 350  | 362  |
|       | 40       | 326        | 336  | 346  | 340      | 349  | 360            | 355      | 366  | 378  |
|       | 95       | 175        | -    | _    | _        | _    | _              | _        | _    | _    |
|       | 90       | 181        | _    | _    | _        | _    | (i—1)          | _        | _    | _    |
|       | 80       | 190        | 197  | _    | 194      | _    | _              |          | -    | _    |
| 21    | 70       | 200        | 207  | 214  | 205      | 214  | _              | 213      | 221  | _    |
|       | 60       | 211        | 219  | 226  | 218      | 226  | 235            | 226      | 235  | 243  |
| ×     | 50       | 221        | 228  | 237  | 230      | 237  | 247            | 239      | 248  | 257  |
|       | 40       | 231        | 238  | 245  | 241      | 250  | 255            | 253      | 262  | 269  |
|       | 95       | 103        | -    | _    | -        | _    | _              | _        | -    | _    |
|       | 90       | 107        | _    | _    | _        | _    | 10-0           | _        | _    | _    |
| 14    | 80       | 112        | 116  | _    | 114      | -    | _              | _        | -    | _    |
|       | 70       | 118        | 122  | 126  | 121      | 126  | 13-3           | 126      | 130  | -    |
|       | 60       | 124        | 129  | 134  | 129      | 134  | 139            | 134      | 139  | 144  |
|       | 50       | 130        | 135  | 140  | 136      | 140  | 146            | 141      | 146  | 152  |
|       | 40       | 136        | 141  | 145  | 143      | 148  | 151            | 150      | 155  | 159  |



# ABANDONING THE AIRCRAFT

The recommended minimum height for abandoning the aircraft is 2000 ft AGL or, in a spin, transition level plus height of ground above sea level

Warn crew
Make RT distress call
Jettison canopy
Release safety harness
Speed as low as possible
Disconnect RT lead
Dive head first towards trailing edge of mainplane

To jettison canopy

Pull down jettison handle Push canopy away if necessary

Note: Above 75 kt it should not be necessary to push the canopy away.

#### DITCHING

If possible, abandon rather than ditch. If ditching inevitable:

Radio ... Transmit distress call

Flap ... ... As required ... Jettison Parachute harness ... Release Safety harness ... Tight

Approach into wind at normal speed with full flap
If power available, hold off just clear of water
Touch down at lowest practicable speed and close throttle
Land on crest of wave if possible or, if the swell is heavy,
along the swell
Aircraft will probably turn on its back
Release harness and leave cockpit

Comms/ Abandoning/ Ditching

# COMMUNICATIONS FAILURE

| UHF | ground  | facilities |
|-----|---------|------------|
|     | availab | le         |

NORM-FAIL switch ... FAIL I/C NORM-EMERG switch EMERG

UHF transmission and reception with intercom using UHF sidetone VHF ground facilities available

NORM-FAIL switch ... FAIL

Receiver switches ... All down

Speak into microphone without operating transmit switch and check if voice audible

VHF receiver switch Up Transmitter selector VHF Then:

Voice audible:
Normal VHF TR available, no intercom
Voice not audible:
Normal VHF receive

Speechless TR available No intercom

# **ELECTRICAL FAILURES**

#### Alternator failure

Indications

Alternator failure warning

light ... ... On

Ammeter ... Negative reading Volts ... Low reading

Actions

Reduce electrical load

Tripped: Alternator switch OFF, reset c/b, then alternator switch on

Not tripped: Alternator switch OFF, then on

If alternator regained, check voltage 29 V max

If alternator *not* regained or voltage above 29 V, switch alternator OFF and trip CHARGE c/b

#### Considerations

Reduced services are available from an 80% charged battery for 45 minutes but satisfactory communications will only be available for 12 minutes (UHF) or 35 minutes (VHF)

For landing, full flap selection and fuel booster pump may not be available

# Flap actuator failure

If flap fails to operate when selected, check flap circuit breaker and if tripped make one attempt to reset

If flap still fails to operate, continue flight with flap in achieved position, observing flap limiting speeds.

Forced Idg/ Propeller/ Electrical

#### FORCED LANDING

The recommended height for abandoning the aircraft is 2000 ft AGL

#### Check:

Fuel booster pump ... OFF
Throttle ... ... Closed
Mixture ... ... CUT-OFF
Ignition ... ... OFF
Fuel selector valve ... OFF
Harness ... ... Tight

Canopy ... ... Closed and locked

Parking brake ... Off

When final flap selection has been made, set battery master switch OFF. Flap operation will be dependent on battery state.

# PROPELLER OVERSPEEDS OR FAILS TO GOVERN

If RPM exceeds 2700 or varies when throttle is moved in the constant speed range:

Set RPM control to minimum Close throttle slowly Reduce speed as much as practicable Open throttle slowly

If governing is restored, reset power
If governing is not restored, close throttle and make a
further attempt to restore

# Persistent failure to govern

Land as soon as possible Maintain lowest practicable speed and power



# Engine Failure in Flight — continued

#### **♦** Non-mechanical Failure

#### Immediate actions

Warn crew

Fuel booster pump ... ON Throttle ... ... Closed

Gain height if possible while reducing speed to 75 kt for the glide

Select landing area, noting wind velocity

Make RT distress call Check altimeter setting

Plan descent

# Subsequent actions

Check for cause of failure Attempt to restart

#### RESTARTING THE ENGINE

WARNING: Do not attempt to restart the engine following either an engine fire or mechanical failure

Attempt to restart the engine by setting or confirming:

Fuel selector valve ... BOTH Fuel booster pump ... ON

Mixture ... Fully rich
RPM ... ... Max
Ignition ... BOTH

Ignition ... ... BOTH
Throttle ... Slightly open

Eng.Failure/ Restarting Press the starter button and release it immediately the engine fires.

Check the starter warning light goes out

OR

If height is available, dive \ until the engine restarts

↓ If stationary, the propeller may not start to windmill below
120 knots

↓

This method of restarting may require 1500 feet

Note: If engine failure was caused by intake icing, use induction hot air until clear of icing conditions.

Forced landing - overleaf

#### FORCED LANDING

The recommended height for abandoning the aircraft is 2000 ft AGL

#### Check:

Fuel booster pump OFF Throttle ... ... Closed ... CUT-OFF Mixture ... ... ... OFF Ignition ... ... Fuel selector valve ... OFF Harness ... ... ... Tight ... Closed and locked Canopy ... ... Parking brake ... Off

When final flap selection has been made, set battery master switch OFF. Flap operation will be dependent on battery state.

# PROPELLER OVERSPEEDS OR FAILS TO GOVERN

If RPM exceeds 2700 or varies when throttle is moved in the constant speed range:

Set RPM control to minimum Close throttle slowly Reduce speed as much as practicable Open throttle slowly

If governing is restored, reset power
If governing is not restored, close throttle and make a
further attempt to restore

# Persistent failure to govern

Land as soon as possible Maintain lowest practicable speed and power



# Engine Failure in Flight — continued

# **♦** Non-mechanical Failure

#### Immediate actions

Warn crew

Fuel booster pump ... ON Throttle ... ... Closed

Gain height if possible while reducing speed to 75 kt for the glide

Select landing area, noting wind velocity

Make RT distress call Check altimeter setting

Plan descent

# Subsequent actions

Check for cause of failure Attempt to restart

# RESTARTING THE ENGINE

WARNING: Do not attempt to restart the engine following either an engine fire or mechanical failure

Attempt to restart the engine by setting or confirming:

Fuel selector valve
Fuel booster pump
Mixture
RPM ... ... Fully rich
Ignition ... ... BOTH

... BOTH

Throttle ... Slightly open

Eng.Failure/ Restarting Press the starter button and release it immediately the engine fires.

Check the starter warning light goes out

OR

↓ If stationary, the propeller may not start to windmill below 120 knots

This method of restarting may require 1500 feet

Note: If engine failure was caused by intake icing, use induction hot air until clear of icing conditions.

Forced landing — overleaf

#### ENGINE FAILURE AFTER TAKE-OFF

#### Immediate actions

◀ Warn crew
Select attitude for gliding speed
Pick a landing area
Lower flap as necessary

# Subsequent actions

Make RT call Carry out forced landing checks (see Card 13)

Note: Circumstances and time available will determine the least hazardous course of action and which of the drills can be completed.

#### ENGINE FAILURE IN FLIGHT

# 

# **Immediate actions**

Warn crew Gain height if possible while reducing speed to 75 kt for the glide

Fuel booster pump ... OFF
Throttle ... ... Closed
Mixture ... ... CUT-OFF
Ignition ... ... OFF
Fuel selector valve ... OFF
Cockpit heating ... Off
Alternator ... OFF

# Subsequent actions

Do not attempt to restart
Select landing area, noting wind velocity
Make RT distress call
Check altimeter setting
Plan descent
Carry out forced landing checks (see Card 13)

#### ENGINE FIRE IN THE AIR

Note: No engine fire extinguisher is fitted.

#### Actions

Warn crew and reduce speed to 75 kt for the glide

Fuel booster pump
Throttle ... Closed
Mixture ... CUT-OFF
Ignition ... OFF
Fuel selector valve
Cockpit heating ... Off
Alternator ... OFF

Make RT distress call

Battery master switch ... OFF when situation permits

If the fire does not go out, abandon if sufficient height is available (2000 ft AGL)

If the fire goes out or height is not sufficient for abandoning, make a forced landing (see Card 13)

Fire

# WARNING: Do not attempt to restart the engine

#### COCKPIT FIRE

Battery master switch ... OFF
Alternator ... OFF
Cabin heating ... Off
Use hand fire extinguisher

Note: If time is sufficient, make RT distress call before battery master switch OFF.

# ENGINE FIRE ON THE GROUND

# Actions

Warn crew
Fuel booster pump ... OFF
Throttle ... ... Closed
Mixture ... ... CUT-OFF
Ignition ... ... OFF
Fuel selector valve ... OFF
Battery master switch ... OFF
Parking brake ... Off

Collect the hand fire extinguisher and vacate the aircraft

Card 11 (AL3)

# BULLDOG T Mk 1 EMERGENCIES

FUMES IN COURPIT

CABIN AIR OFF SELECT SURGEN DEMIST

DEMIST

BELOW 120 KB

OPEN CANOPY 2 Nothing

RT CAM

RTB

Set Medic.

' ENGINE FIRE ON THE GROUND/ ENGINE FIRE IN THE AIR/ COCKPIT FIRE

ENGINE FAILURE/ RESTARTING THE ENGINE

FORCED LANDING/ PROPELLER MALFUNCTIONING/ ELECTRICAL FAILURES

COMMUNICATIONS FAILURE/ ABANDONING/ DITCHING